

MARS GLOBAL SURVEYOR

AEROBRAKING OPERATIONS DECISION PROCESS

(AODP)

M. D. JOHNSTON

MDJ-1 6/28/96



INTRODUCTION

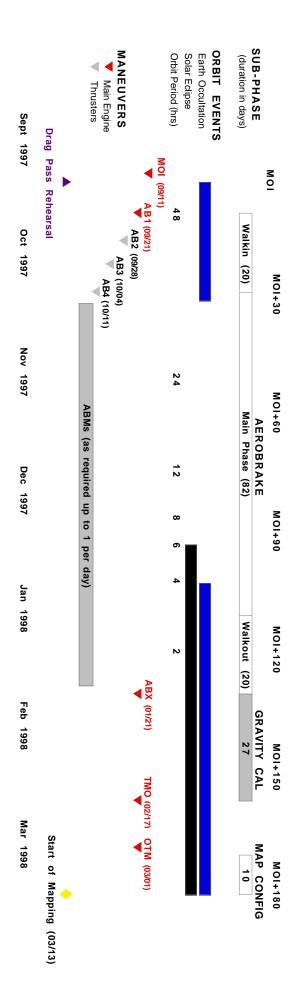
PROVIDES A FUNDAMENTAL FRAMEWORK TO FACILITATE DURING AEROBRAKING FLIGHT OPERATIONS PLANNING AND DECISION MAKING

=> SYSTEMS LEVEL VIEW OF AEROBRAKING OPERATIONS

- BASED ON THE USAGE OF THE MISSION OPERATIONS SYSTEM (MOS) CORE PROCESSES
- PARALLEL AND SERIAL ACTIVITIES DECISION MAKING - SHOWS THE MOS PROCESS INTERACTION AS FLOW NECESSARY TO SUPPORT AEROBRAKING OPERATIONS ILLUSTRATES (AT THE PROCESS LEVEL) THE DATA PATHS AND
- IDENTIFIES KEY DECISION JUNCTURES THAT LEAD EITHER TO THE AEROBRAKING DELAY NEXT AEROBRAKING SUB-PHASE OR A PATH THAT MAY FORCE AN

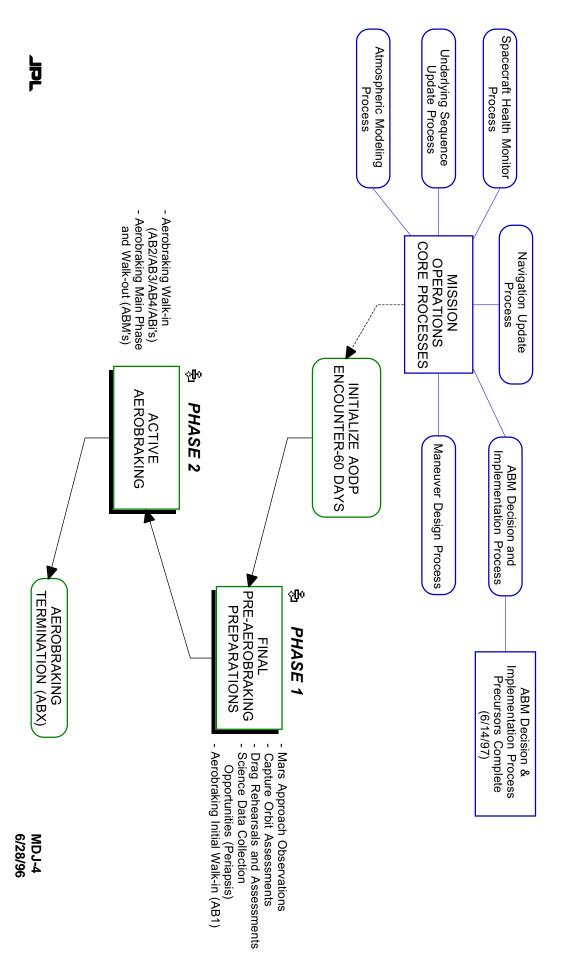


ORBIT INSERTION PHASE TIMELINE (LAUNCH PERIOD OPEN)



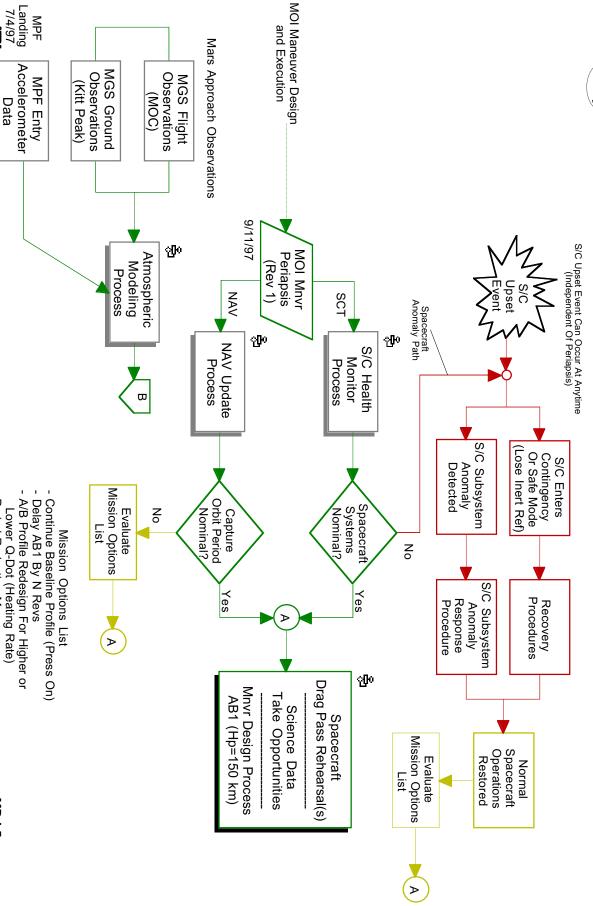


=> MARS SURVEYOR OPERATIONS PROJECT (MSOP) FLIGHT OPERATIONS MANAGER <= AEROBRAKING OPERATIONS DECISION PROCESS PROCESS OWNER



Mars Global

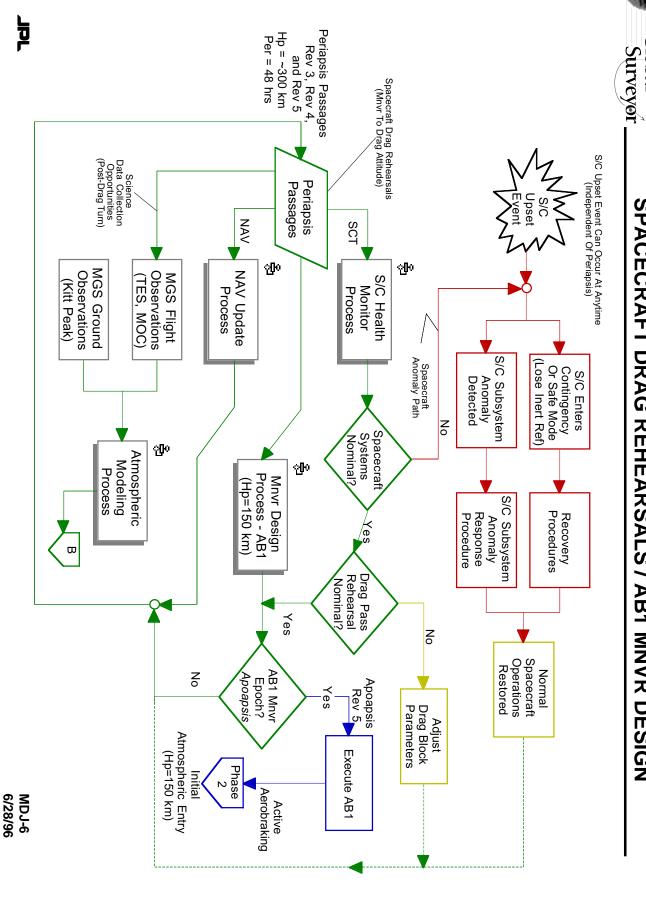
MARS APPROACH OBSERVATIONS / CAPTURE ORBIT ASSESSMENTS AODP PHASE 1 - FINAL AEROBRAKING PREPARATIONS



Period Reduction Maneuver

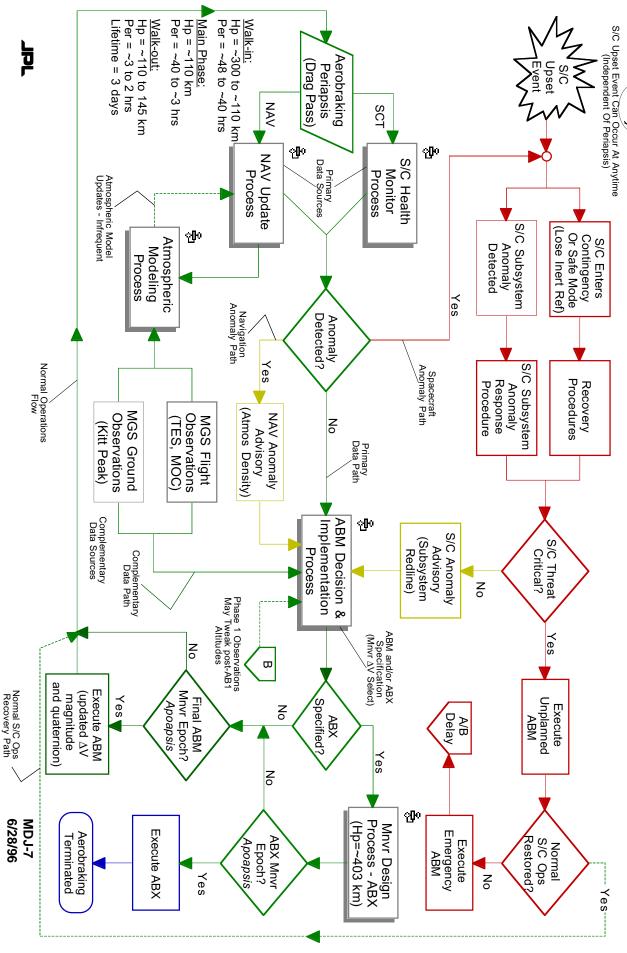
MDJ-5 6/28/96

Mars AODP PHASE 1 - FINAL PRE-AEROBRAKING PREPARATIONS **SPACECRAFT DRAG REHEARSALS / AB1 MNVR DESIGN**



Mars Global Surveyor_

AODP PHASE 2 - ACTIVE AEROBRAKING WALK-IN, MAIN PHASE, WALK-OUT





MISSION OPERATIONS CORE PROCESSES (1 OF 2)

- SPACECRAFT HEALTH MONITOR PROCESS
- TO MONITOR SPACECRAFT HEALTH AND PREDICT SPACECRAFT PERFORMANCE
- NAVIGATION UPDATE PROCESS
- TO GENERATED TRIGGER COMMANDS FOR SPACECRAFT DRAG PASS SCRIPTS AND TO GENERATE SPACECRAFT EPHEMERIDES
- UNDERLYING SEQUENCE UPDATE PROCESS
- TO UPDATE SPACECRAFT SCRIPT INTERNAL TIMING DUE TO ORBIT PERIOD CHANGES AND TO COMMAND SCIENCE ACTIVITIES
- MANEUVER DESIGN PROCESS
- TO DESIGN AND IMPLEMENT THE CRUISE TCM'S (TRAJECTORY MANEUVERS (I.E. MOI, AB1, ABX, TMO, ETC.) CORRECTION MANEUVERS) AS WELL AS OTHER SELECT PROPULSIVE



MISSION OPERATIONS CORE PROCESSES (2 OF 2)

- ABM (AEROBRAKING TRIM MANEUVER) DECISION AND IMPLEMENTATION PROCESS
- NEXT ABM MANEUVER FOR CORRIDOR CONTROL MAINTENANCE POST-AB1 (AB2/AB3/AB4/ABi) AND TO SELECT AND IMPLEMENT THE TO SELECT AND IMPLEMENT THE AEROBRAKING WALK-IN MANEUVERS
- ATMOSPHERIC MODELING PROCESS
- ATMOSPHERIC BEHAVIOR ATMOSPHERIC CONDITIONS AND TO BETTER UNDERSTAND TO UPDATE THE ATMOSPHERIC MODEL TO REFLECT CHANGING



PHASE 1 - FINAL PRE-AEROBRAKING PREPARATIONS

MARS APPROACH OBSERVATIONS

PURPOSE: OBSERVE MARTIAN ATMOSPHERE DURING APPROACH

CHARACTERIZED BY: NOMINAL FLIGHT OPERATIONS COUPLED WITH THE INITIATION OF FLIGHT AND GROUND BASED OBSERVATIONS

TERMINATED BY: MOI MANEUVER

CAPTURE ORBIT ASSESSMENTS

PURPOSE: DETERMINE CAPTURE ORBIT AND ASSESS SPACECRAFT STATE AS A RESULT OF THE MOI MANEUVER

CHARACTERIZED BY: NOMINAL FLIGHT OPERATIONS

TERMINATED BY: INITIATION OF THE SPACECRAFT DRAG REHEARSALS

SPACECRAFT DRAG REHEARSALS / INITIAL WALK-IN (AB1 MNVR DESIGN)

PURPOSE: FINAL PREPARATIONS FOR AEROBRAKING

CHARACTERIZED BY: SPACECRAFT DRAG REHEARSAL AND ASSESSMENTS (MANEUVER SPACECRAFT TO THE AEROBRAKING DRAG ATTITUDE) AND SCIENCE DATA COLLECTION OPPORTUNITIES AT PERIAPSIS

TERMINATED BY: AB1 EXECUTION



PHASE 2 - ACTIVE AEROBRAKING

- WALK-IN (AB2/AB3/AB4/ABi)

PURPOSE: CHARACTERIZE THE MARTIAN ATMOSPHERE DURING LATE

SEPTEMBER / EARLY OCTOBER 1997

CHARACTERIZED BY: ESTABLISHMENT OF MARTIAN ATMOSPHERIC CONDITIONS THROUGH MULTIPLE REDUCTIONS IN PERIAPSIS ALTITUDE

TERMINATION CONDITION: ESTABLISHMENT OF MAIN PHASE NOMINAL HEATING RATES - Q-DOT_{MAIN} = 0.38 W/CM^2

MAIN PHASE (ABM'S)

PURPOSE: LARGE SCALE REDUCTION OF THE ORBIT ENERGY

CHARACTERIZED BY: MAXIMUM NOMINAL SPACECRAFT HEATING RATES, DRIFTS TOWARD THE NORTH POLE ($\omega = \sim 100.0 \text{ deg at LP OPEN}$) INFREQUENT Rp LOWER (ABM) MANEUVERS, ARGUMENT OF PERIAPSIS

TERMINATED BY: ESTABLISHMENT OF A 3-DAY ORBIT LIFE TIME

WALK-OUT (ABM'S)

PURPOSE: CONTINUED REDUCTION OF THE ORBIT ENERGY UNTIL PROPULSIVE MEANS CAN ESTABLISH THE MAPPING ORBIT

CHARACTERIZED BY: MAINTENANCE OF 3-DAY ORBIT LIFE TIME, REDUCTION IN DRIFTS SOUTH MANEUVERS, ARGUMENT OF PERIAPSIS CROSSES THE NORTH POLE AND THE SPACECRAFT NOMINAL HEATING RATES, DAILY RP RAISE (ABM)

TERMINATED BY: APOAPASIS ALTITUDE = 450 KM

DESCENDING NODE TIME = 2:00 PM



RESPONSE TO MDT AI #41 (1 OF 2)

- AEROBRAKING OPERATIONS DECISION PROCESS (AODP) AND MDT AI #41 - CLARIFY FLIGHT TEAM INVOLVEMENT IN THE DEFINE TEAM INPUT PRODUCTS FOR THE AODP
- AODP HAS BEEN UPDATED -
- REVISITED THE OVERALL AODP PROCESS FLOW
- DATA PATHS AND DATA FLOW ARE CLEARLY DELINEATED
- OPERATIONS PLANNING HAVE BEEN CLEARLY IDENTIFIED DECISION JUNCTURES NECESSARY TO SUPPORT FLIGHT
- INTERCHANGE OF TEAM PRODUCTS AND THE ASSOCIATED CDR - BROOKS MOS PRESENTATION - 9/25/95 (NO CHANGE) DESIGN PROCESS WERE PRESENTED AT THE MISSION SYSTEM PROCESS, 2) THE NAV UPDATE PROCESS, AND 3) THE MANEUVER DELIVERY FREQUENCY FOR 1) THE SPACECRAFT HEALTH MONITOR
- FORCED AEROBRAKING CONTROL PARAMETERS INTO THE ABM LEVEL) DECISION AND IMPLEMENTATION PROCESS (REMOVED FROM TOP



RESPONSE TO MDT AI #41 (2 OF 2)

RECOMMENDATION IS MADE TO CLOSE AI #41 WITH THE **FOLLOWING PROVISOS:**

- A) DETAILS OF THE ATMOSPHERIC MODELING PROCESS NEED TO BE WORKSHOP - 6/18/96): WORKED (IDENTIFIED AT 2ND MGS MARS ATMOSPHERIC
- 1) FLIGHT OPS ATMOSPHERIC MODEL UPDATE SUPPORT AND
- 2) ABM DECISION AND IMPLEMENTATION PROCESS SUPPORT
- B) DETAILS OF THE AEROBRAKING CONTROL PARAMETERS WORKED (QUANTIFIED AND DOCUMENTED - APG OPS PROCEDURE) ABM DECISION AND IMPLEMENTATION PROCESS NEED TO BE (SPACECRAFT SUBSYSTEM AND TRAJECTORY) CONTAINED IN THE
- AEROBRAKING WALK-IN SUB-PHASE
- ABi (i=2,n) TARGET ALTITUDES AND MANEUVER FREQUENCY (RESPONSE TO SELECTION OF THE NEXT ABI ALTITUDE STEP) ATMOSPHERIC SENSING (DENSITY OBSERVATIONS) AND STRATEGY FOR THE
- AEROBRAKING MAIN AND WALK-OUT SUB-PHASES
- CORRIDOR CONTROL LIMITS
- HPLOW SPACECRAFT DYNAMIC PRESSURE AND HEATING LIMITS
- HPHIGH "GLIDE SLOPE" (TRAJECTORY) PROFILE NECESSARY TO ACHIEVE 2:00 PMMDJ-13



- REQUIREMENTS FOR INITIATION OF AEROBRAKING OPERATIONS
- NOMINAL FLIGHT PROFILE (AEROBRAKING REFERENCE TRAJECTORY) ACTUAL LAUNCH DATE CONSISTENT WITH THE SPACECRAFT OPERATING ENVELOPE AND
- ABM DECISION & IMPLEMENTATION PROCESS PRECURSORS
- SPECIFICATION OF AV MAGNITUDE SETS
- FUNCTION OF EXPECTED ARGUMENT OF PERIAPSIS MOVEMENT DURING **AEROBRAKING**

SPECIFICATION OF QUATERNION LOOK-UP TABLE PARAMETERS

ABM CHARACTERIZATION

PLANNED ABM MANEUVER (ADVANCED OR FINAL) - NORMAL OPERATIONS

ADVANCED: MNVR EPOCH > 48 HOURS FROM EXECUTE

MNVR EPOCH < 48 HOURS FROM EXECUTE

=> GENERATE S/C TRIGGER COMMANDS AND UPLINK

UNPLANNED ABM MANEUVER - RESPONSE TO A SPACECRAFT ANOMALY

=> AEROBRAKING IS SLOWED

EMERGENCY ABM MANEUVER - RESPONSE TO A MAJOR SPACECRAFT ANOMALY => AEROBRAKING IS DELAYED (POP-UP MANEUVER)



SAMPLE AV MAGNITUDE AND QUATERNION LOOK-UP TABLE

~Five ΔV Magnitude Sets Necessary Walk-in Main Phase - Late

Main Phase - Early Walk-out

"Off-the-Shelf" Emergency Mnvr (Single Size)

One "Up" Quaternion Look-up Table and One "Down" Quaternion Look-up Table

				1.0	(1/2 X) $(1X)$ $(2X)$	Half Size Full Size Double Size	∆V Magnitude Set (m/s)
270.0			140.0	150.0	Periapsis (deg)	Arg Of	Up Table (Raise Periapsis)
qu				q			
q1 q2 q3 q4				q1	(Quaternion Table)	Direction	
q2				q1 q2 q3 q4			
q3				q3			
q4	•	•		q4			
						_	
				0.5	(1/2 X)	Half Size	۸ ۸۷
				0.5 1.0	$(1/2 X) \qquad (1X)$	lalf Size Full Size	∆V Magnitude Set
				1.0			∆V Magnitude Set (m/s)
270.0			140.0	1.0	(1X)	Full Size	
270.0 dn			140.0 .	1.0 2.0	(1X) (2X)	Full Size Double Size	
dn			140.0	1.0 2.0 150.0 dn	(1X) (2X) Periapsis (deg)	Full Size Double Size Arg Of	
dn			140.0	1.0 2.0 150.0 dn	(1X) (2X)	Full Size Double Size	∆V Magnitude Set (m/s) Down Table (Lower Periapsis)
270.0 dn q1 q2 q3 q4			140.0	1.0 2.0	(1X) (2X) Periapsis (deg)	Full Size Double Size Arg Of	